

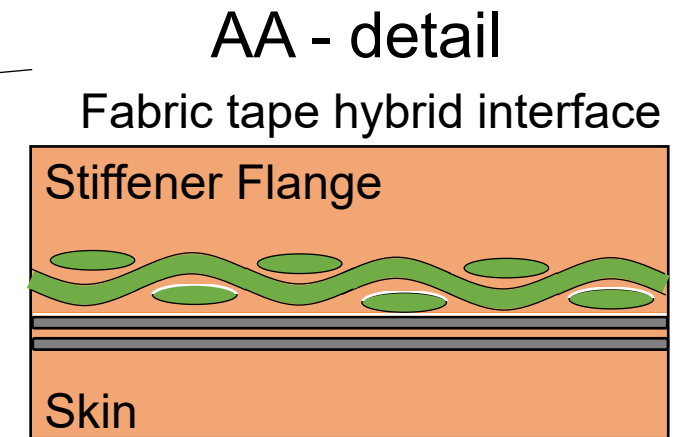
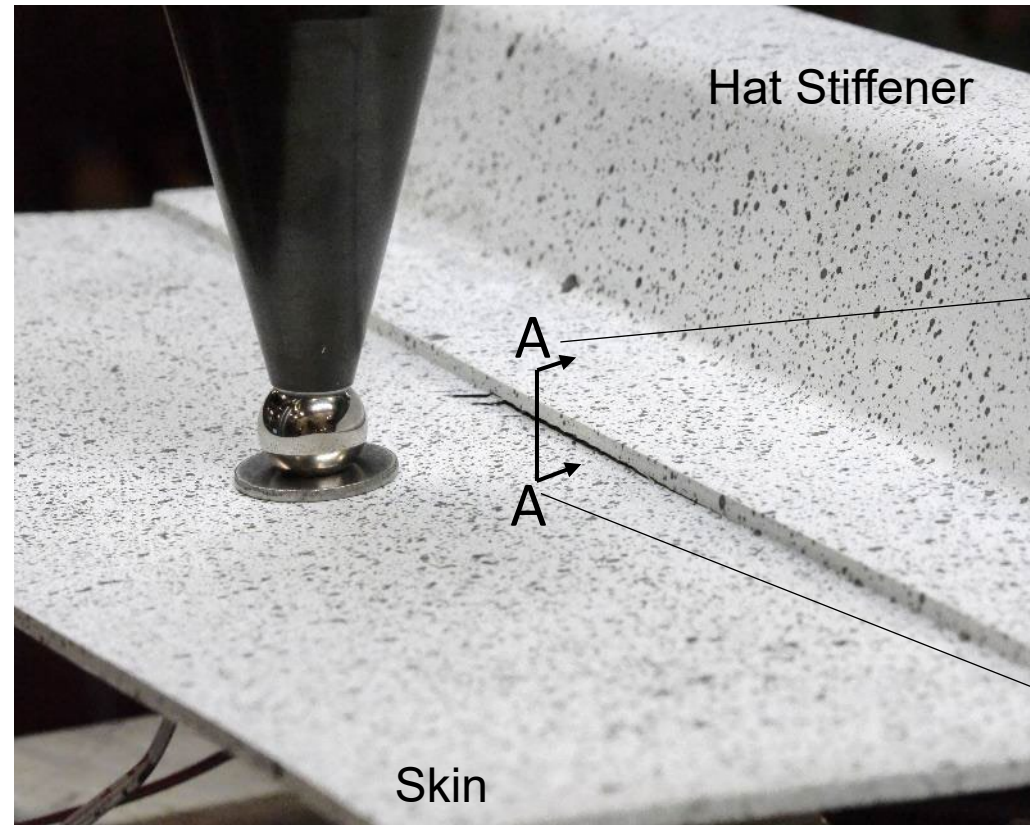
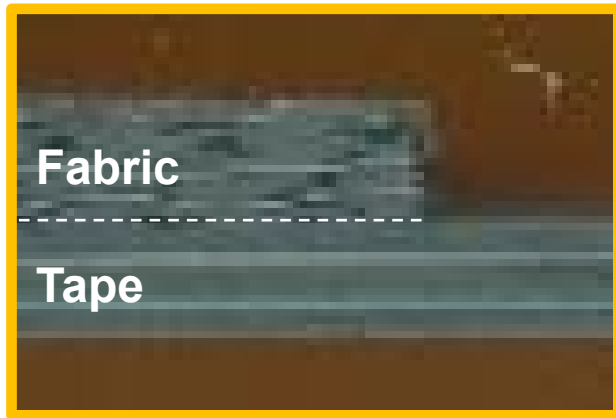
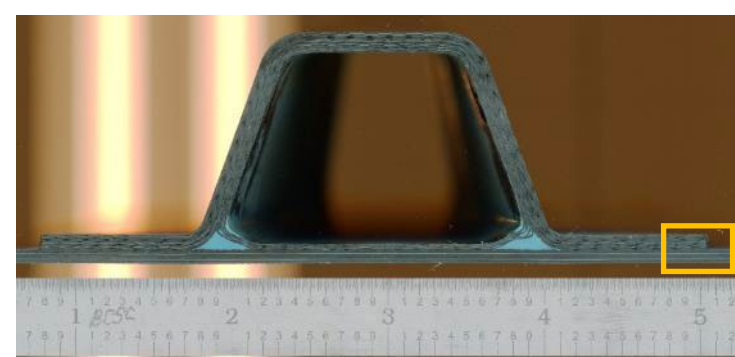


Mode-II Interlaminar Fracture Toughness and Factors Affecting it

James Ratcliffe and Nelson de Vieira Carvalho
Durability, Damage Tolerance, and Reliability Branch
NASA Langley Research Center, Hampton, VA

Polymer Matrix Composite (PMC) Laminates

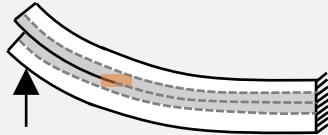
- Fabric plies often added to tape PMC laminates for protection and ease of manufacturing



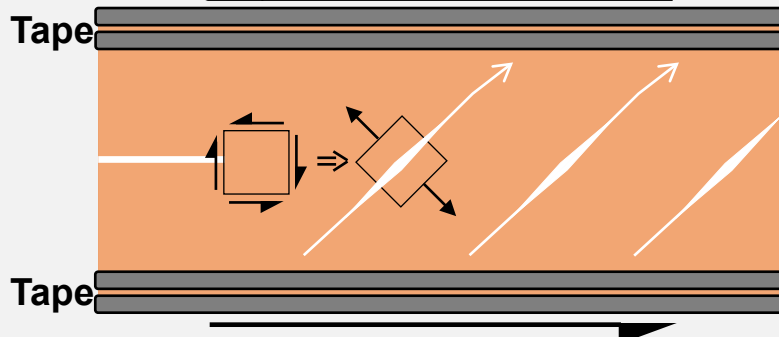
Interlaminar Fracture (Delamination) in PMC Laminates

Tape laminates

Mode-II



Tape fibers prevent intralaminar crack growth

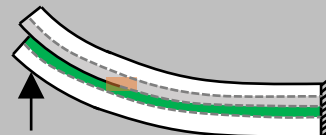


Unique interlaminar fracture toughness:

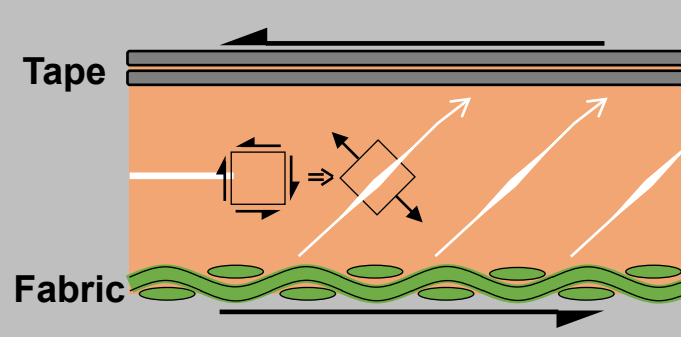
$$G_{IIC} - \text{Tape}$$

Hybrid tape/fabric laminates

Mode-II: Tape ply skimmed



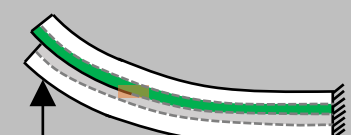
Tape fibers prevent intralaminar crack growth



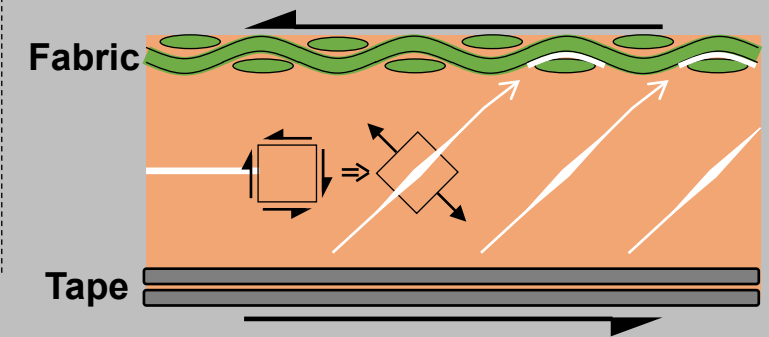
Fracture toughness not unique?

$$G_{IIC} - \text{Tape/Fabric}$$

Mode-II: Fabric ply skimmed



Crack paths both sides of fabric weft tows?



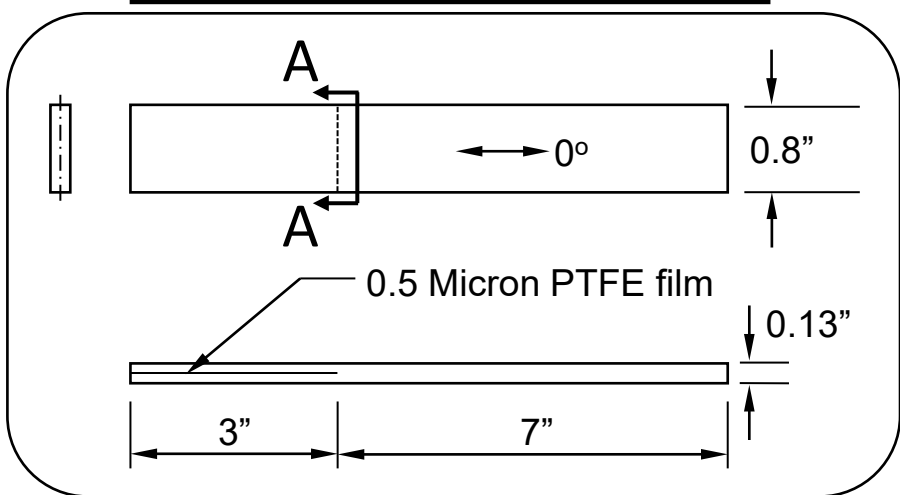
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$$G_{IIC} - \text{Fabric/Tape}$$

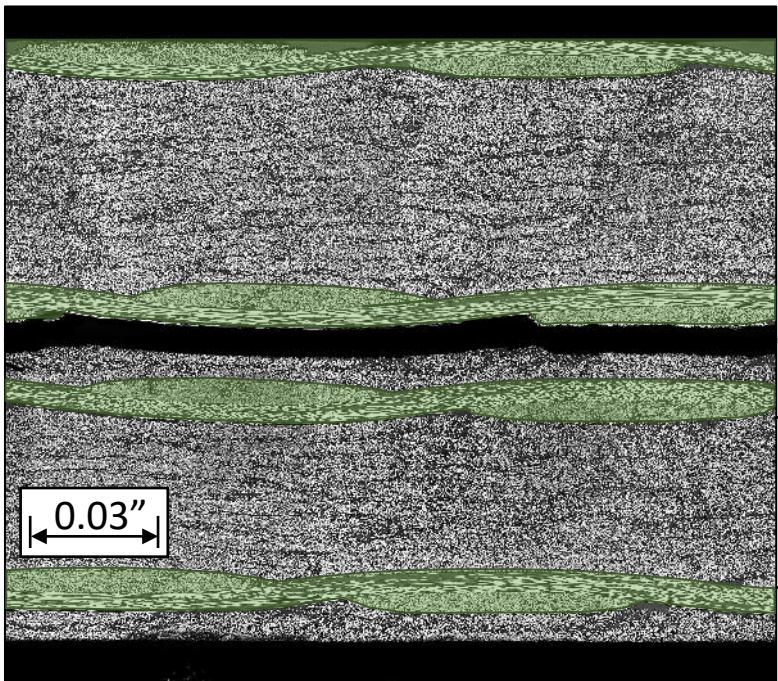
Materials and Interlaminar Fracture Specimens

- 18-ply laminates:
IM7/8552 carbon/epoxy unidirectional tape and plain weave fabric plies
- Layup with minimal mechanical coupling and thermal residual stresses
- Starter delamination at tape/fabric ply interface

Interlaminar Fracture Specimen

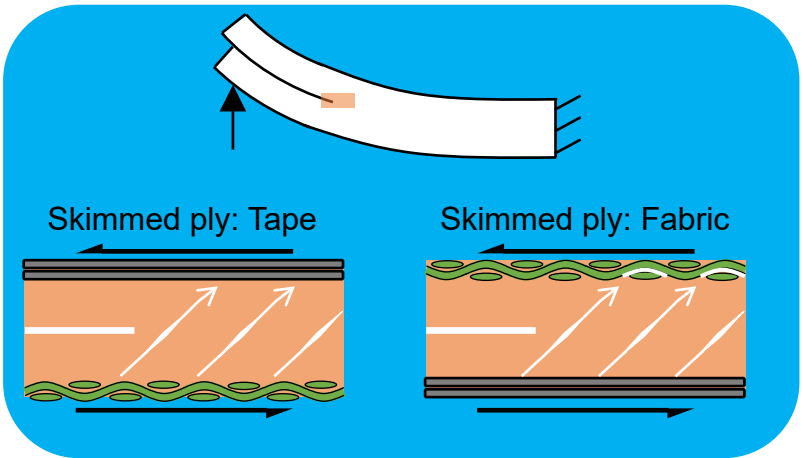


AA - detail



Stacking Sequence

Ply Type	Orientation
Fabric	0
Tape	0
Tape	0
Tape	0
Tape	0
Tape	0
Tape	0
Tape	0
Fabric	0
PTFE film	-
Tape	0
Fabric	0
Tape	0
Tape	0
Tape	0
Tape	0
Tape	0
Fabric	0
Tape	0



Mode-II Fracture Toughness, G_{IIc}

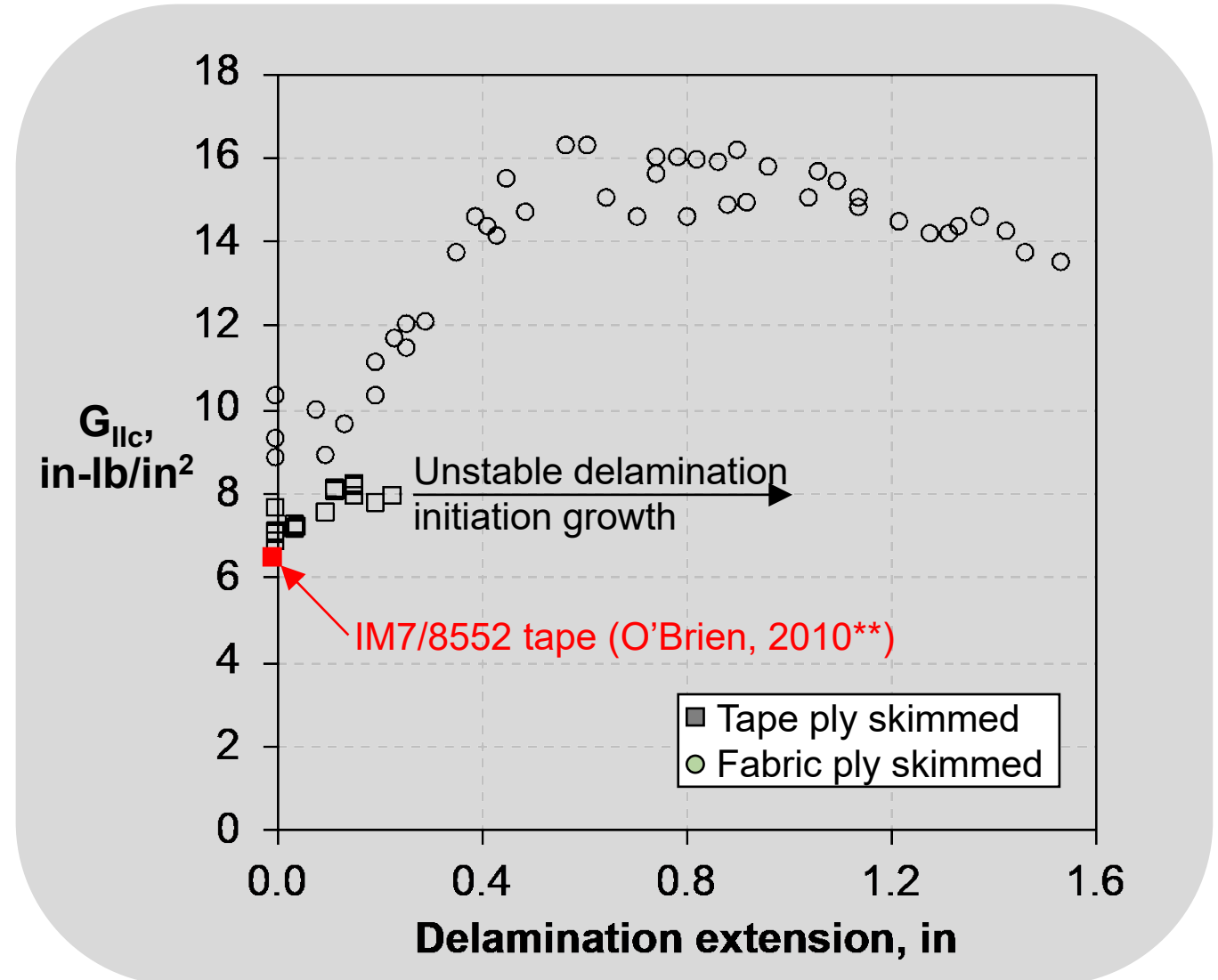
- Tape skimmed specimens:
Constant G_{IIc} prior to unstable growth
- Fabric skimmed specimens:
Significant increase in G_{IIc} with delamination extension

Initiation G_{IIc} , in-lbf/in ²	
Tape	Fabric
7.8 (10.1)*	9.5 (15.1)

Propagation G_{IIc} , in-lbf/in ²	
Tape [#]	Fabric ^{&}
7.9 (8.8)	15.1 (3.4)

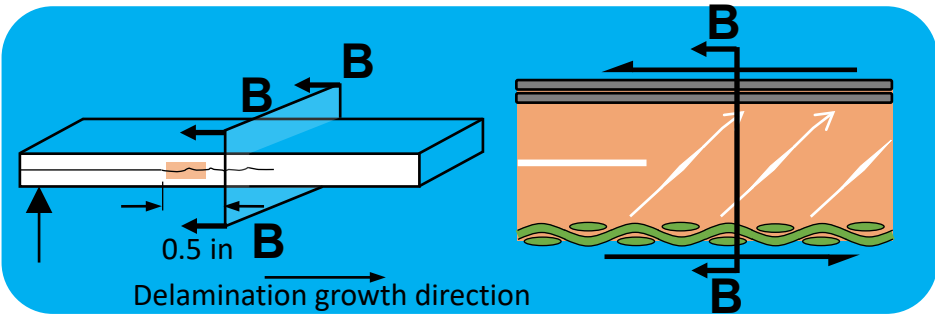
* Coefficient of variation (%) in parenthesis

Average of all propagation G_{IIc} values from tape containment specimens
& Average of propagation G_{IIc} values for delamination extension greater than 0.5" from fabric containment specimens

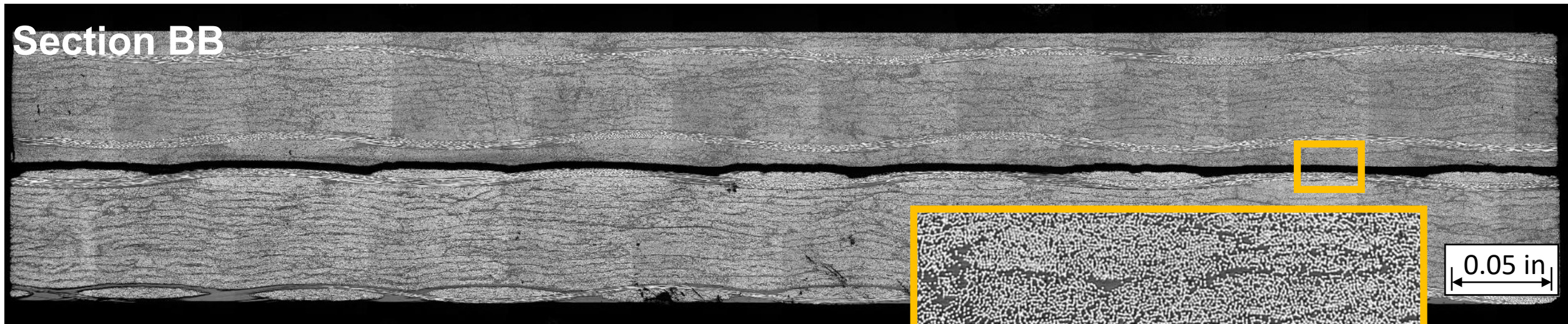


**O'Brien, T.K, et al, "Mode II Interlaminar Fracture Toughness and Fatigue Characterization of a Graphite Epoxy Composite Material." NASA/TM-2010-216838, Aug. 2010 (Table 2, pp. 16)

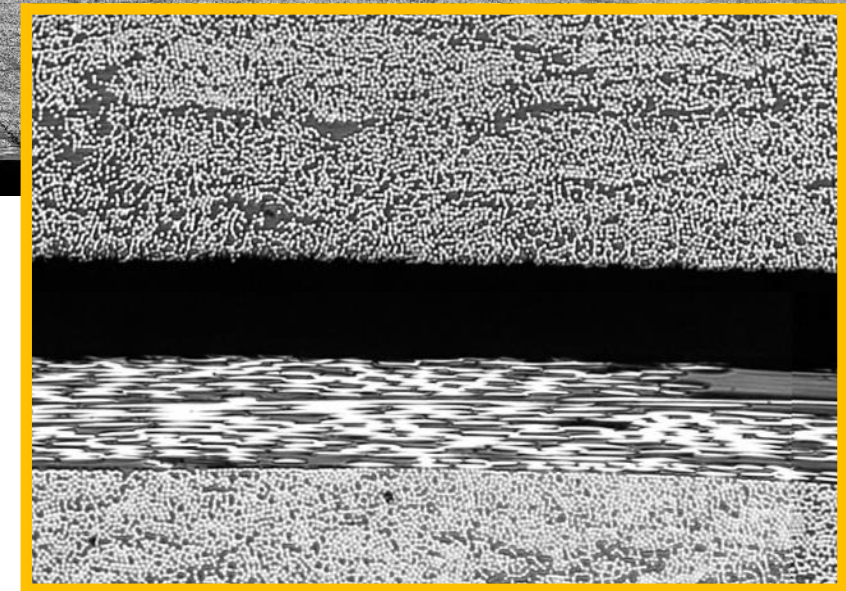
Tape Skimmed Specimen Cross-Section



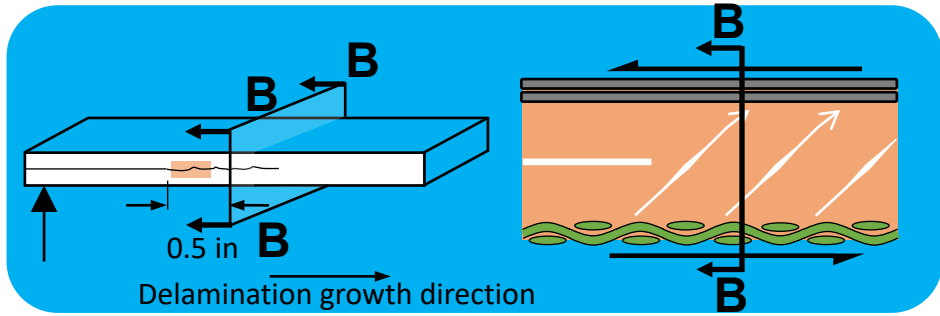
BB – 0.5 in ahead of starter delamination front



Delamination contained within tape/fabric ply interface

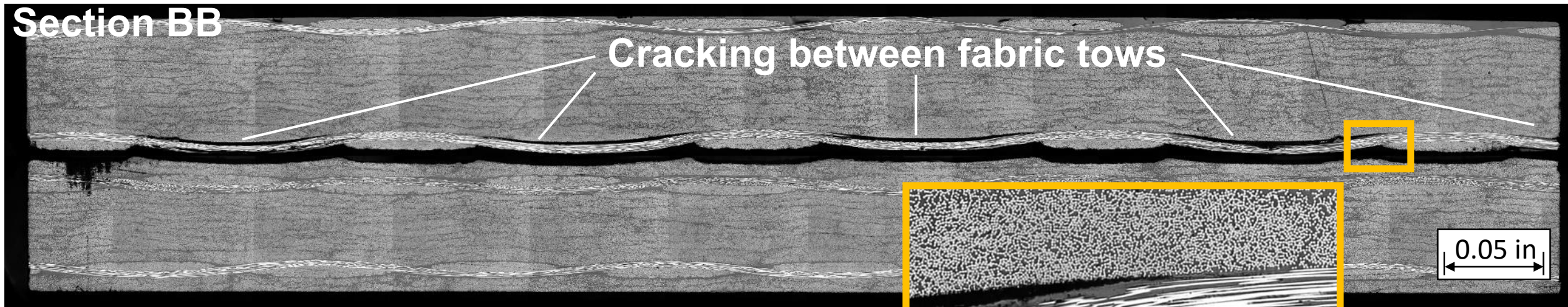


Fabric Skimmed Specimen Cross-Section

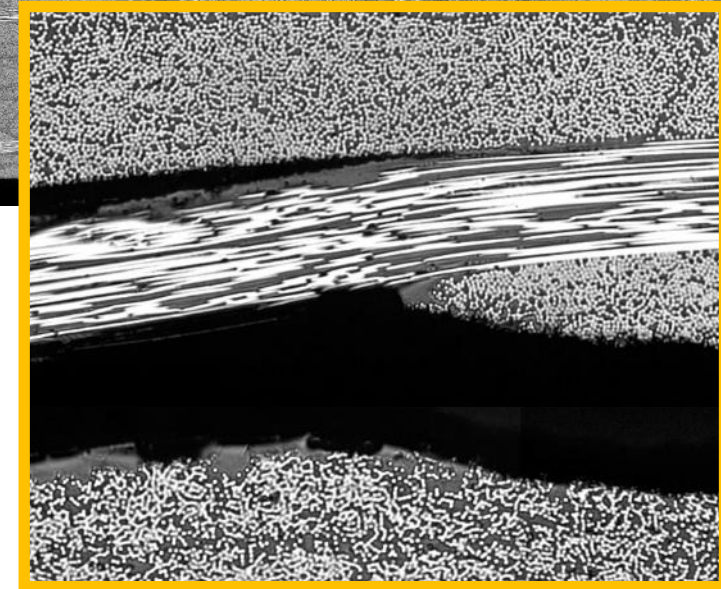


BB – 0.5 in ahead of starter delamination front

Section BB



Cracking occurs between warp and weft tows causing delamination resistance

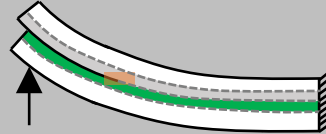


Closing Remarks:

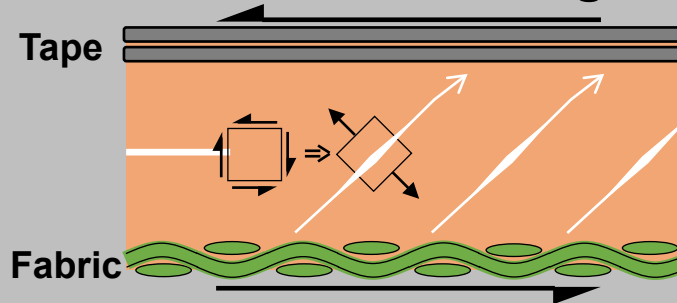
- Hybrid tape/fabric fracture specimen developed
- Delamination skimming fabric plies results in greater resistance
- Hybrid interfaces exhibit different fracture toughness as a function of skimmed ply
- Key for design and simulation

Hybrid tape/fabric laminates

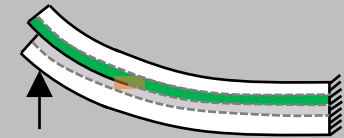
Mode-II: Tape skimmed



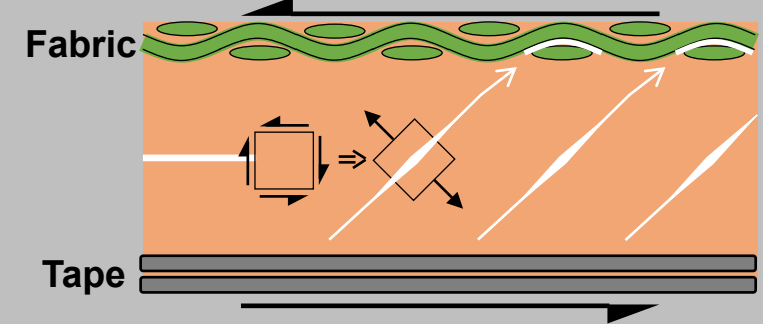
Tape fibers prevent intralaminar crack growth



Mode-II: Fabric skimmed



Crack paths both sides of fabric weft tows



Fracture toughness not unique

$G_{IIc} - \underline{\text{Tape}}$ /Fabric

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$G_{IIc} - \underline{\text{Fabric}}$ /Tape